

MODIFIED METHOD OF GRAVITY MODEL APPLICATION FOR TRANSATLANTIC AIR TRANSPORTATION

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Abstract: Air transportation between Europe and the U.S. is becoming more and more significant. It can only hardly be replaced by other means of transportation, since its biggest advantages include speed and reliability. Air transportation forecasting is important for planning the development of airports and related infrastructure, and of course also for air carriers. Therefore, it is important to forecast the number of flights between selected airports in Europe and the U.S. and the number of transported persons. A gravity model is usually used for this forecasting. Determination of coefficients which significantly affect results of the formulas used in the gravity model is crucial. Coefficients are, as a rule, computed by an iterative algorithm implementing the gradient method. This technique has some limitations if the state space is inappropriate. Moreover, the exponent parameter in the formula is obviously fixed. We have chosen the new method of differential evolution to determine the gravity model coefficient. Differential evolution works with populations similarly to other evolution algorithms. It is suitable for solving complex numerical problems. The suggested methodology can be helpful for various airlines to forecast demand and plan new long-haul routes.

Key words: Air transportation, development, forecasting, gravity model, transatlantic flights, differential evolution

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1. Introduction

Aviation-related planning and services need to continuously be improved so that operations can be more efficient and profitable. Commercial air transportation is important in connecting people and businesses in the world.

Transatlantic long-haul flights fly regularly from Europe to North America, South America, the Far East and Australia, Africa and vice versa, already since the beginning of commercial aviation in 1939.

This paper describes the possibilities of current data analysis methodology, which can help identify the main factors that influence passenger demand for the transatlantic route network.

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Passengers obviously demand airlines operating on the most direct routes, with comfortable aircrafts and inexpensive tickets. The required reliability and safety of flights is also gradually on the rise. Airlines should follow these trends and try to enhance passenger's experience and achieve a high level of passenger satisfaction.

Aviation as a specific branch of industry is very dynamic. Positive external factors include, for example, technological development, development of infrastructure and Air Traffic Management (ATM). Negative factors include, for example, capacity limits, airport congestion, oil crisis, global airline deregulation, terrorist attacks and political instability.

Route forecasting is one of the numerous decision support tools created by airlines and represents a critical part of profitable network planning. Especially the identification and forecasting of a new market and associated revenue can potentially lead to an increase of the profit of the respective airline. This could lead to increased passenger demand from the new market. Airline route decisions can be made simply based on an individual's judgment based on experience, but this becomes more difficult when the number of routes as well as the size of the airline increase. Then it is necessary to choose an appropriate model to forecast demand and plan new routes.

The number of passengers which travel between airports in a specific time interval is usually estimated using the gravity model. The parameters (also referred to as constants) of this model are frequently calculated with iterative algorithms implementing the gradient method.

The gravity model predicts movement of persons, information and goods between cities, or even between continents. Hence these models measure intensity of relations between 2 objects (small relations between small objects, large relations between large ones). In terms of traffic network management, the gravity models are used for managing the impact of technical and economic parameters of the individual traffic network sections.

In air transportation forecasting, there exists a gravity model for analyzing passenger demand, described by Chang [2]. The formula is:

$$T_{ij} = \frac{C \cdot P_i \cdot A_j}{f(d_{ij})},\tag{1}$$

where:

- T_{ij} represents the number of trips produced in country *i* (origin) and attracted to country *j* (destination) as the force between the masses,
- C is a constant,
- P_i represents the production factors of country i,
- A_j represents the attracting factors of country j,
- d_{ij} is the distance between country *i* and country *j*.

This article uses a modified gravity model with Cheu's gravity model formula [3], in which the Eq. (1) is rewritten into the specific form:

$$T_{ij} = a_i b_j \frac{\text{POP}_i \cdot \text{BUS}_j}{\text{DIS}_{ij}^x},\tag{2}$$

where:

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- POP_i is the population of origin airport i,
- BUS_i is the business or attractiveness of destination airport j,
- DIS_{ij} is the distance between the airport of origin *i* and the destination *j* and it is used as the impedance of travel between *i* and *j*,
- T_{ij} is the predicted number of flights between origin airport *i* and destination *j*,
- a_i is an airport specific trip production constant,
- b_i is an airport specific trip attraction constant.
- a_i, b_j and the exponent x are parameters of the model and they need to be calibrated.

The initial value x is set to 2 and parameters a_i, b_j are computed via an iterative algorithm [3, 12], as follows:

Assume all
$$a_i = 1$$
,
calculate all b_i using $b_j = \left[\sum_{i=1}^n \frac{a_i \text{POP}_i}{\text{DIS}_{ij}^x}\right]^{-1}$, (3)

calculate all
$$a_i$$
 using $a_i = \left[\sum_{j=1}^n \frac{b_j \text{BUS}_j}{\text{DIS}_{ij}^x}\right]^{-1}$, (4)

calculate
$$T_{ij} = a_i b_j \frac{\text{POP}_i \text{BUS}_j}{\text{DIS}_{ij}^x}$$
 for all *i* and *j*. (5)

Steps (3), (4) and (5) are repeated until a_i , b_j and T_{ij} converge. The result of the application of a gravity model is a demand matrix \mathbf{T} (T_{ij} is one of elements of this matrix). In this paper, parameters used in gravity model are determined using the method of differential evolution.

Differential evolution is a relatively new type of evolutionary algorithms (since 1994). There are many algorithms that are classified as evolutionary, for example Genetic Algorithms, Ant Colony Optimization, Scatter Search, Immunology System Method. Each algorithm is suitable for solving a certain class of problems. Hence it is important to test and decide for which set of problems the algorithm is applicable and which approach fits best to the respective task.

The algorithm of differential evolution is demonstrated on the real flight prediction problem applied to six European airports as origins and six U.S. airports as destinations.

2. Differential evolution

Differential evolution (DE) is one of many evolutionary techniques. It was created by R. Storn and K.V. Prince in the nineties [6, 12]. It is suitable for numerical optimisation problems. Basic advantages of differential evolution are:

– simplicity,

- heterogeneity of representation (integers, float numbers),

- running time of the algorithm,
- independence of parent quality three parents are selected for reproduction randomly [6, 9],
- good ability to find an extreme,
- efficiency of non-linear problem solving with boundaries.

2.1 Description of the DE algorithm

Similarly to other evolutions algorithms, DE works with populations, where a population is a set of individuals in simulation time t and each individual is one solution of the problem. The principle of population creation and parameter setting is similar to other evolutionary techniques. The population can be represented as an $NP \times D$ matrix, where NP is population size and D is the dimension (the number of parameters of the individual).

In other words, columns of the matrix are individuals, and each individual is a one-dimensional vector having D components. Each individual within the population is marked as J_i ($i = 1 \dots NP$). The quality of each individual is estimated by the calculation of the fitness function, which measures also the individual's suitability for subsequent evolution (the result of the fitness function is called CV – cost value).

Parameters and terminology

The algorithm of differential evolution is parameterized by the following parameters:

- Crossover threshold CR the probability of noisy vector selection. Each component of the testing vector is set to the component of a noisy vector with probability CR and to the component of the parent r_4 otherwise. The recommended value is CR $\in \langle 0.8 0.9 \rangle$ [6].
- The size of the population (the count of individuals in the population) NP, with a recommended value of $NP \in \langle 2D, 100D \rangle$ where D is the dimension of the individual.
- Mutation constant F is used as multiplicative constant in the noisy vector creation process. The recommended value is $F \in \langle 0, 2 \rangle$ [6].
- Maximum count of iterations MAX_ITERATION the number of evolution cycles after which the algorithm stops.

The algorithm can be described as follows:

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Generate initial generation randomly
iteration = 1;
while (iteration <MAX_ITERATION)
{</pre>
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Apply reproduction cycle;
Store best individual (solution);
Iteration = iteration+1;
}
```

Reproduction cycle

The individual is represented as a vector with D components. Denote each component as $x_j = 1 \dots D$; x_{ij} is the *j*-th component of the *i*-th individual.

A noisy vector is a vector $\mathbf{v} = (v_1, \ldots, v_D)$ and the "testing" vector is $\mathbf{z}_{\mathbf{v}} = (z_{v1}, \ldots, z_{vD})$.

For each individual J_i in population (i = 1 ... NP). { Select randomly 4 parents r_1, r_2, r_3, r_4 from the population other than J_i // noisy vector **v** creation for j = 1 to D do $v_j = x_{r3,j} + F(x_{r1,j} - x_{r2,j})$;

// testing vector $\mathbf{z}_{\mathbf{v}}$ creation for j = 1 to D do if (rand() < CR) $z_{vj} = v_j$; else $z_{vj} = x_{r4,j}$;

//calculate a fitness value CV of testing vector
fit = fitness($\mathbf{z}_{\mathbf{v}}$);
if (fit($\mathbf{z}_{\mathbf{v}}$) is better than fit(J_i))
copy $\mathbf{z}_{\mathbf{v}}$ into interpopulation into *i* position;
else
copy original individual J_i into interpopulation into *i* position
}
Replace population by interpopulation.

Note: Function rand () generates a random value between 0 and 1 with uniform distribution.

2.2 Application of differential evolution to the specific issue

DE was tested on a task with six departing airports in Europe and six destination airports in the U.S.

2.2.1 Airport selection

The process of selecting airports for future development of demand forecasting methodology was conducted by a qualitative method. First, the judgment method was used to investigate six airports in the U.S.: New York (JFK), Boston (BOS), Chicago (ORD), Miami (MIA), Los Angeles (LAX) and San Francisco (SFO). A few airports in Europe that have competing flights with Prague and passengers from Prague usually take connecting flights from Prague via these cities to the

U.S.: Vienna, Frankfurt, Paris, London, Amsterdam, Copenhagen, and Zurich, supported by market analysis from Prague Airport, worldwide statistics and current situation of PRG.

IATA code	European Airports
AMS	Amsterdam Airport Schiphol
BUS	Budapest Liszt Ferenc Inter. Airport
CDG	Charles De Gaulle Airport
FRA	Frankfurt am Main Airport
LHR	Heathrow Airport
PRG	Vaclav Havel Airport Prague
VIE	Vienna International Airport
ZRH	Zurich Airport

Tab.	Ι	Final	result	of	sei	lection	of	^r Euro	pean	air_{i}	ports.
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IATA code	European Airports
BOS	Logan International Airport
JFK	John F. Kennedy International Airport
LAX	Los Angeles International Airport
MIA	Miami International Airport
ORD	Chicago O'Hare International Airport
SFO	San Francisco International Airport

Tab. II Final result of selection of U.S. airports.

2.2.2 Input data

There are two types of variables:

- Dependent (predicted) variable is the number of passengers (passenger trips) on an airport-pair route during a set period (typically).
- Independent variables are related mainly to two factors [11]:
 - Geo-economic factors / economic activity,
 - Geographical factors / location impacts.

Based on availability of historical data and previous research [2], ten variables are statistically significant in determining passenger flows between airport pairs (Tab. VI).

Data collection – demand data, supply data (airline schedules), economical (or geo-economical) data.

Demand data were obtained based on the total number of passengers travelling from Prague Airport to the various U.S. airports in 2011, and the same also applies

Code	Dependent variables	Description				
PAX	Number of passengers	Number, one-way from European				
		to U.S. airports				
Code	Independent variables	Description				
POP	Population	Number, in 2011				
UER	Unemployment rate	% , in 2011				
DIS	Distance	Distance between two selecte				
		cities, in air miles				
TPAX	Total passengers of the airport	Total passengers handled in 2011				
NIPC	National income	US\$ (per capita) of the destination				
		city in 2011				
BUS	Business	Total number of companies in the				
		destination city in 2007				

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Tab. III Defined variables.



Fig. 1 Part of the process of estimation of passenger demand from the supply data. Data was mainly collected from available sources on the Internet, such as US DOT, U.S. Department of Commerce, Airports Council International or ACI, Boeing, ICAO, IATA and so on. The rest had to be estimated based on statistics, predictions and by indexing.

to geo-economic data. Supply data were collected to fill up the missing demand data that were not obtainable, especially for trips originating from European airports other than Prague Airport.

For the purpose of this research, the supply data of the direct flights was obtained from Expedia [13] and collected in two distinct periods: the summer

timetable (August 2012) and winter timetable (March 2013) of the airlines. For each flight between a defined airport pair, the following data were collected on a daily basis (for one week in the defined period): operated airlines, co-shared airlines, operated aircraft and their capacity based on configuration of the seats.

The biggest challenge of this research was the collection and estimation of the annual passenger demand from the detected European airports to the selected U.S. airports. All existing scheduled flights between defined airport-pairs had to be collected from available flight databases, an extra sheet in Excel was created for each airport-pair and passenger demand was estimated based on load factors (operated aircraft and flight frequencies).

After completing a database of required data, index methods were used on the core database of input data for further simulation. Additionally, the required statistics data such as population, unemployment rate, total annual passengers of the origin and destination airports, national income and number of companies was collected for future modelling and demand estimation.

BUS_i	(no.of companies)	POP_j	(population)
BOS	49 667	AMS	780 559
JFK	944 129	CDG	$11 \ 800 \ 000$
LAX	450 108	CPH	$1\ 213\ 822$
MIA	85 146	FRA	691 518
ORD	255 502	LHR	$14 \ 900 \ 000$
SFO	$105 \ 030$	ZRH	390 082

Specific values are provided in tables bellow:

Tab. IV Parameters BUS_i , POP_j . Source: [12].

D (mi)	BOS	JFK	LAX	MIA	ORD	SFO
AMS	$3\ 450$	3 630	5560	999 999	4 110	5 460
CDG	3 440	3 620	$5\ 650$	4 580	4 140	$5\ 570$
CPH	999 999	999 999	999 999	999 999	4 260	999 999
FRA	$3\ 660$	3 840	5 790	4 820	4 330	5680
LHR	$3\ 250$	3 440	$5\ 440$	4 410	3 940	$5\ 350$
ZRH	3 730	3 920	5 920	4 870	4 4 30	5 820

Tab. V The distance matrix D (elements DIS_{ij}) as gravity model input. Source: [12].

2.2.3 Searching parameters of gravity model using DE

DE is used to determine the following parameters of the gravity model: coefficients a_i and b_j , $i = 1 \dots n$, $j = 1 \dots m$ and exponent x in Cheu's formula [3], where:

- n is the number of rows in DIS (PAX) = number of origins airport,
- m is the number of columns in DIS (PAX) = destination cities.

Р	BOS	JFK	LAX	MIA	ORD	SFO	Q_{i_pax}
AMS	89 122	$300 \ 857$	$122 \ 187$	0,001	102 908	59 937	$675 \ 012$
CDG	72 724	573 733	$275 \ 051$	$139\ 624$	$150 \ 323$	180 817	$1 \ 392 \ 271$
CPH	0,001	0,001	0,001	0,001	69 867	0,001	69 867
FRA	80 266	401 663	94 956	82 361	245 730	204 670	$1 \ 109 \ 646$
LHR	414 737	$1\ 500\ 386$	$357 \ 313$	$247 \ 136$	$521 \ 369$	$255 \ 257$	$3 \ 296 \ 198$
ZRH	65 191	203 929	54 665	$100 \ 697$	68 727	63 776	$556 \ 985$
D_{j_pax}	722 040	$2\ 980\ 569$	904 172	$569\ 818$	$1\ 158\ 923$	$764 \ 457$	7 099 979

Tab. VI The matrix **P** of estimated number of passengers (elements PAX_{ij}). Source: [12].

The individual is a vector with dimension equal to D = n + m + 1, where the meaning of components is: the first *n* components are a_i coefficients, the next *m* components are b_j coefficients, and the last component is the exponent *x*.

The structure of an individual is: length D = n + m + 1

a_1	a_2	 a_n	b_1	b_2	 b_m	x

Fig. 2 Structure of chromosomes.

If the initial population is generated, then random values are usually generated within a given interval in DE. When a reproduction cycle is executed, new values of the testing vector are tested for membership in the permissible interval (this tests if they are valid; the boundaries are based for example on physical constrains). In our application, the initial individuals are generated randomly with boundaries which are derived from recommended initial values by the iteration method used in gravity model:

x is generated from the range of $x \in \langle 1, 3 \rangle$,

 a_i, b_j are generated from the range of $a_i, b_j \in \langle 0, 3 \rangle$.

During reproduction cycles, the test is not performed to verify if the DE can find optimal parameters outside of recommended range in the classical gravity model.

Fitness function calculation

An estimation of matrix **T** is calculated for each individual using parameters a_i, b_j , x. The known **P** matrix is divided by the value 200 (average number of passengers per flight) and the number of flights is calculated. The fitness function is total square root error:

$$FIT = \sum_{i=1}^{n} \sum_{j=1}^{m} (T_{ij} - PAX_{ij}/200)^2.$$
 (6)

DE searches for a minimum value of the fitness function.

Parameters of algorithms:

Algorithm parameters are as follows: Size of population: N = 1,000. Maximum number of iterations MAX_ITERATIONS = 10.000 – the algorithm stops if the count of iterations is 10,000.

Crossing threshold CR = 0.8; the value is derived from the recommended value in the literature [6, 9].

Mutation constant F = 1.

The final solution is the best solution over all generations.

3. Results of DE

Results are provided for the data presented above (Tab. VI).

Original results

DE found the following optimal parameters of the gravity model:

 $\begin{array}{l} a_1 = 3362.88\\ a_2 = 418.21\\ a_3 = 138.32\\ a_4 = 6899.30\\ a_5 = 679.16\\ a_6 = 6296.69\\ b_1 = 2964.49\\ b_2 = 777.79\\ b_3 = 2991.06\\ b_4 = 4221.19\\ b_5 = 1798.43\\ b_6 = 9359.23\\ x = 4.24 \end{array}$

This table contains input data for the fitness function (6) – distance matrix PAX_{ij} (Tab. V) divided by 200.

Р	BOS	JFK	LAX	MIA	ORD	SFO
AMS	446	1504	611	0	515	300
CDG	364	2869	1375	698	752	904
CPH	0	0	0	0	349	0
FRA	401	2008	475	412	1229	1023
LHR	2074	7502	1787	1236	2607	1276
ZRH	326	1020	273	503	344	319

Tab. VII Input data – matrix P (elements $PAX_{ij}/200$).

This is a demand matrix T_{ij} as a side result of DE. The matrix contains predicted numbers of flights calculated by (5) using parameters $a_1, \ldots, a_6, b_1, \ldots, b_6, x$ evolved by the DE.

RD SFO	69434.0 $363.217036.0$	141256.0 $627.460079.0$	77688.0 22.750850.0	57379.0 $558.329970.0$	399727.0 $1526.509654.0$	00526.0 $259.237219.0$
MIA O	265.809749.0 566.3	526.162856.0 1032.4	13.641762.0 30.77	409.604629.0 825.1	266.706466.0 2611.8	201.843094.0 385.6
LAX	460.597987.0	808.919807.0	28.064511.0	704.930358.0	1947.858400.0 1	330.296263.0
JFK	1532.757125.0	2915.530101.0	75.881141.0	2194.656799.0	7422.475281.0	1035.245910.0
BOS	381.315244.0	725.761400.0	18.742504.0	539.422402.0	1893.803104.0	256.265360.0
E	AMS	CDG	CPH	FRA	LHR	ZRH

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SFO	63.532036	-276.624921	22.750845	-465.020030	250.224654	-59.642781
ORD	51.829434	280.826256	-318.557312	-403.492621	5.054727	41.965526
MIA	265.809744	-171.957144	13.641757	-2.200371	31.026466	-301.641906
\mathbf{LAX}	-150.337013	-566.335193	28.064506	230.150358	161.293400	56.971263
JFK	28.472125	46.865101	75.881136	186.341799	-79.454719	15.600910
BOS	-64.294756	362.141400	18.742499	138.092402	-179.881896	-69.689640
	AMS	CDG	CPH	FRA	LHR	ZRH

Tab. IX Absolute differences $T_{ij} - PAX_{ij}/200$.

The absolute differences between predicted numbers of flights T_{ij} and input data $PAX_{ij}/200$ are provided in Tab. IX.

Relative differences between predicted number of flights T_{ij} and input data $PAX_{ij}/200$ are calculated by:

$$R_{diff} = \frac{T_{ij} - PAX_{ij}/200}{PAX_{ij}/200} \cdot 100[\%].$$
(7)

	BOS	JFK	LAX	MIA	ORD	SFO
AMS	-14.4	1.9	-24.6	5316194889.2	10.1	21.2
CDG	99.6	1.6	-41.2	-24.6	37.4	-30.6
CPH	374849975.5	1517622728.7	561290114.6	272835137.9	-91.2	455016898.4
FRA	34.4	9.3	48.5	-0.5	-32.8	-45.4
LHR	-8.7	-1.1	9.0	2.5	0.2	19.6
ZRH	-21.4	1.5	20.8	-59.9	12.2	-18.7

Tab. X Relative differences in %.

The input data for flights between the Copenhagen Airport (CPH) and the airports in Boston (BOS), New York (JFK), Los Angeles (LAX), Miami (MIA), and San Francisco (SFO), and for flights between Amsterdam Airport (AMS) and Miami Airport (MIA) were not obtained and thus the outputs are not relevant. In Tab. VII, this is indicated with "o".

The results provided in Tab. X (relative differences in %) for the airports specified above are not valid. They are presented here only for informational purposes. Graph at Fig. 3 shows how the fitness function (total square root error) evolved.



population/generation

Fig. 3 Evolution of the fitness function.

4. Conclusion

The results obtained using the DE methods are more credible. Take, for example, the number of flights between Amsterdam (AMS) and New York (JFK), which is otherwise stated to amount to 100,416 per year, which represents 275 flights per day, and this number appears to be unrealistic.

Parameters obtained		Parameters obtained				
by the DE		by	the iterative algorithm [12]			
a_1	3362.88	a_1	16.147			
a_2	418.21	a_2	15.462			
a_3	138.32	a_3	111.577			
a_4	6899.30	a_4	16.046			
a_5	679.16	a_5	14.897			
a_6	6296.69	a_6	16.318			
b_1	2964.49	b_1	0.000002			
b_2	777.79	b_2	0.000002			
b_3	2991.06	b_3	0.000002			
b_4	4221.19	b_4	0.000002			
b_5	1798.43	b_5	0.000001			
b ₆	9359.23	b_6	0.000002			
Х	4.24	х	2			

Tab. XI Parameters obtained by the DE and by the iterative algorithm.

The above-stated calculations demonstrate how the values of constants a_i and b_j , $i = 1 \dots n$, $j = 1 \dots m$ and exponent x can be calculated using the differential evolution method. These constants depend on a number of aspects of the given location and it is the method of their calculation using differential analysis that enables their calibration using the data entered.

Entered data were adopted from [12]. Presented results obviously differ from those published in [12] as DE is used instead of an iterative algorithm. Comparison can show that the predicted demand matrix \mathbf{T} calculated with parameters searched via DE is more accurate.

If we compare the modified results with values that have been calculated without using the method of differential evolution, it is evident that they vary considerably. However, the original results have not yet been used in practice for the forecast of development of air transportation between specified destinations. The method of differential evolution may bring new knowledge for processing of air traffic forecasts.

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	BOS	JFK	LAX	MIA	ORD	SFO
AMS	381.315244.0	1532.757125.0	460.597987.0	265.809749.0	566.369434.0	363.217036.0
CDG	725.761400.0	2915.530101.0	808.919807.0	526.162856.0	1032.441256.0	627.460079.0
CPH	18.742504.0	75.881141.0	28.064511.0	13.641762.0	30.777688.0	22.750850.0
FRA	539.422402.0	2194.656799.0	704.930358.0	409.604629.0	825.157379.0	558.329970.0
LHR	1893.803104.0	7422.475281.0	1947.858400.0	1266.706466.0	2611.899727.0	1526.509654.0
ZRH	256.265360.0	1035.245910.0	330.296263.0	201.843094.0	385.600526.0	259.237219.0

Tab. XII Assumed number of flights $-\mathbf{T}$ matrix as a result of the DE.

		·	<u> </u>	·						
	POP_i	187086	2832190	245076	165969	3576031	93626	7099979		
	A_i	16.1478	15.4626	111.5768	16.0459	14.8966	16.3175			7099979
2nd iteration $(k = 2)$	SFO	11391	162287	1895	9716	203888	5466	0.00000	2	394643
	ORD	21161	304545	220948	17868	385446	10064	0.00000	1	960032
	MIA	126	135773	1355	7926	170240	4509	0.00000	2	319930
	LAX	48722	696292	8222	41523	873091	23400	0.00000	2	1691251
	JFK	100416	1456824	12049	84511	1846016	47686	0.00000	2	3547502
	BOS	5270	76468	607	4425	97349	2500	0.00000	2	186620
	T_{ij}	AMS	CDG	CPH	FRA	LHR	ZRH	b_j		BUS_{j}

Tab. XIII Number of flights – **T** matrix as a result of iterative algorithm Source: [12].

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